

Velikost sprejemnika - HeNe laser

$$A = v_0^2 \cdot \pi = A_s = v^2 \cdot \Omega$$

$$D_0 = \frac{4\pi}{\lambda^2} \cdot A =$$

$$D_0 = \frac{4\pi}{\lambda} \Rightarrow \Omega = \frac{4\pi}{D_0} = [\text{sr}]$$

TV satelit - smernost, sprejeta moč, slabljenje, najnižja fr.

$$A_g = \int v^2 \Omega_{\text{satelit}}$$

$$D_0 = \frac{4\pi}{\lambda} = \frac{4\pi v^2}{A_g} = \frac{4\pi v^2}{\pi v_0^2} [\text{dbi}]$$

$$P_s = \frac{P_0 A_s \eta_s \eta_0}{A_g} = \frac{P_0 \pi v_s^2 \eta_s \eta_0}{\pi v_0^2} [\text{dBm}]$$

$$a = 10 \log \frac{P_s}{P_0} [\text{dBm}]$$

$$A_0 = \pi v_0^2 = \frac{\lambda^2}{4\pi} D_0 \rightarrow \lambda^2 = \frac{4\pi v_0^2}{D_0}$$

$$\lambda = \frac{2\pi v_0}{\sqrt{D_0}} \quad f \geq \frac{c_0}{\lambda}$$

TV sateliti - razmak?

$$D_s = \frac{4\pi}{\lambda^2} A_s = \frac{4\pi}{\lambda^2} \cdot \pi v_s^2$$

$$\Omega = \frac{4\pi}{D_s} \quad h = r - r \cos \alpha$$

$$A = v^2 \cdot \Omega = 2\pi r v h = 2\pi r^2 v (1 - \cos \alpha)$$

$$\Omega = 2\pi (1 - \cos \alpha) \Rightarrow \cos \alpha = 1 - \frac{\Omega}{2\pi}$$

Razvoj v vrsto: $\cos \alpha = 1 - \frac{\alpha^2}{2} + \dots$

$$\frac{\alpha^2}{2} \approx \frac{\Omega}{2\pi} \quad \alpha \approx \sqrt{\frac{\Omega}{\pi}} \Rightarrow \text{razmak } \times Z!$$

Satelit do mobilnega upov. oddajna moč?

$$P_s = \frac{U_s^2 Z_{\text{eff}}}{Z} =$$

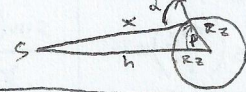
$$A_s = \frac{\lambda^2}{4\pi} G_s = \frac{G_s}{4\pi} \left(\frac{c_0}{f}\right)^2$$

$$A_g = \pi v^2$$

$$P_0 = P_s \cdot \frac{A_g}{A_s}$$

↳ razmerje površin (slabi)

TV satelit - oddajna moč?



$$x = \sqrt{(h + R_z)^2 + R_z^2 - 2(h + R_z)R_z \cdot \cos \alpha}$$

$$\frac{x}{\sin \alpha} = \frac{h + R_z}{\sin(180^\circ - \alpha)} = \frac{h + R_z}{\sin \alpha}$$

$$\alpha = \arcsin\left(\frac{h + R_z}{x} \sin \alpha\right)$$

Geog. področje manjše zaradi ukrivlj. zemlje: $A_g' = A_g \cdot \cos \alpha$

$$P_0 = S A_g' \cdot \cos \alpha$$

Najmanjša razdalja med antenama

$$\lambda = \frac{c_0}{f}$$

$$v_0 \geq \frac{Z d^2}{\lambda} \leftarrow \text{Fraunhofer pogoji}$$

$$v \geq v_{0,\text{min}} + v_{s,\text{min}}$$

$$v \geq v_0 \cdot Z$$

Max fazna napaka

$0 \leq \alpha < \pi/2$	$\frac{\pi d^2}{8\lambda}$
$\pi/2 \leq \alpha < \pi$	$\frac{\pi d^2}{4\lambda}$
$\pi \leq \alpha < 3\pi/2$	$\frac{\pi d^2}{2\lambda}$
$3\pi/2 \leq \alpha < 2\pi$	$\frac{\pi d^2}{\lambda}$

Dolžina lijaka za sprejem SAT signala

$$A = \frac{\lambda^2}{4\pi} \cdot D \rightarrow \text{dbi?}$$

$$\lambda = \frac{a^2 + b^2}{2\lambda}$$

Globoko zrcalo uporabno od f_{min}

$$Z_r = 5\lambda \Rightarrow$$

$$\Rightarrow \lambda = \frac{Z_r}{5}$$

$$f = \frac{c_0}{\lambda}$$

Parabolno zrcalo dovoljena odstopanja smernosti 1dB in 0.25dB

$$\lambda = \frac{c_0}{f}$$

$$D_{-1\text{dB}} \leq \frac{\lambda}{16}$$

$$D_{-0.25\text{dB}} \leq \frac{\lambda}{32}$$

Razmerje $\frac{f}{d}$, gorisče v odprtini $f = h$

$$f = \frac{d^2}{16h} \cdot f$$

$$\frac{f^2}{d^2} \approx \frac{f}{16h}$$

$$f^2 = \frac{d^2 \cdot f}{16h}$$

$$\frac{f}{d} = \sqrt{\frac{f}{16h}} = \frac{1}{4} \sqrt{\frac{f}{h}}$$

Da sat. ubcži iz tirnice (motor vključimo v perigeju)

$$a_1 = \frac{1}{2} (h_a + h_p + 2R_z) =$$

$$a_2 = \infty$$

$$r_p = h_p + R_z$$

$$v_1 \left. \begin{matrix} v_2 \end{matrix} \right\} = \sqrt{\mu \left(\frac{2}{r_p} - \frac{1}{a_1} \right)} \quad \Delta v = v_2 - v_1$$

Življenjska doba satelita s popravki

$$\Delta v_c = v_i \ln \frac{m}{m - m_0}$$

$$t = \frac{\Delta v_c}{\frac{\Delta v}{\Delta t}}$$

Višota vseh Δv za prevoz geost. sat.

$$v_z = v_p = R_z \quad a_g =$$

$$v_g = \sqrt{\frac{\mu}{a_g}} \quad a_p = \frac{r_a + r_p}{2}$$

$$\Delta v_1 = \sqrt{\mu \left(\frac{2}{r_p} - \frac{1}{a} \right)} - v_z$$

$$\Delta v_2 = v_g - \sqrt{\frac{\mu}{r_a} - \frac{1}{a}}$$

$$\Delta v = \Delta v_1 + \Delta v_2$$

Satelit - odstopanje položaja na nebu

$$d = (R_z + h) \cdot 2 \sin\left(\frac{\Omega \lambda \cos \alpha}{2}\right)$$

TV satelit - dobitok spr. ant.

$$P_0 G_0 = \text{dBW}$$

$$\lambda = \frac{c_0}{f} \quad P_s = P_0 G_0 G_s \left(\frac{\lambda}{4\pi d}\right)^2$$

$$G_s = \frac{P_s}{P_0 G_0} \left(\frac{4\pi d}{\lambda}\right)^2$$

Satelit - premer zrcala sprejemne antene

$$P_s G_0$$

$$P_s = P_0 G_0 \frac{A_{\text{eff}}}{4\pi d^2} \rightarrow A_{\text{eff}} = \frac{P_s}{P_0 G_0} 4\pi d^2$$

$$A_{\text{eff}} = \eta \pi r^2 \rightarrow Z_r = Z \sqrt{\frac{A_{\text{eff}}}{\eta \pi}}$$

Moč sat odd. $P_0 Z$

$$P_s = \frac{U_s^2}{Z} \quad A_g = \frac{\lambda^2}{4\pi} G_s =$$

$$= \frac{G_s}{4\pi} \left(\frac{c_0}{f}\right)^2$$

$$A_g = \pi v^2$$

$$P_0 = P_s \cdot \frac{A_s}{A_g}$$